

Free Vibration Response Of Double-Bay Multi-Storey Building Frames With Stiffened Joints

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ABSTRACT

This paper investigated self-excited vibration response of double-bay multi-storey building frames for the effect of joint stiffening on natural frequencies. One of the frames has normal rigid joints. Three others of the frames have stiffened joints of stiffened lengths: 250mm, 400mm and 750mm respectively. On account of limitations of shear frame models, the frames are modeled as frames with flexible horizontal members having multi degrees of freedom (MDOF). Classical displacement method of analysis is adopted using stiffness coefficients which are modified to include and embrace the contributions of joint stiffening. Results revealed: joint stiffening increased the natural frequencies of the frame; natural frequencies also increased with the length of stiffening. In addition, joint stiffening brought about enhanced values of other dynamic characteristics of the frame which are functions of natural frequencies.

Keywords: Free Vibration; Stiffening factor;
Natural frequencies; Restoring forces;
Multi-degree of freedom (MDOF);
Rotational and translational stiffness coefficients.

1. INTRODUCTION

Frames are the main load bearing components of structures of building and other skeletal formations such as ship, bicycle, bridge to mention but a few. They are made up of straight separate vertical, horizontal or inclined members connected together by means of welded, bolted, riveted or other types of joints.

Type and strength of the connecting joints are key elements to the overall stability of the frame. It is for this reason rigid types of connection are given preference over the hinged or pinned joints [1]. Furthermore, [2 and 3] have shown that stiffening the joints reduced the stress as well as enhanced the stability and the performance of the structure.

Frame can be a plane or space structure. Plane frames are two-dimensional and stable only in their own plane whereas space frames are three-dimensional and stable in all directions [4,6].

Forces acting on the frame structures are, very often, dynamic in nature. These time-dependent forces are characterized by variable intensity, frequency, and state, plane of acting and sense or direction, with respect to time. Most forms of loading have dynamic components, and some forms of structures especially where they are slender, are susceptible to the dynamic effects. Furthermore, the use of the structure, say as a laboratory housing sensitive instruments, may require that vibration be considered. Thus, there is an interrelation between the sources of dynamic excitation, the structural form and the purpose of the structure [5].

A structure [6] is said to be undergoing free vibration when it is disturbed from its position of static equilibrium and then allowed to vibrate without any external dynamic excitation. By free vibration, it is meant the motion of a structure without any dynamic excitation, external loads or forces or support motion. Free vibration is initiated by disturbing the structure from its equilibrium position by some initial displacements and/or by imparting some initial velocities [5, 6].

Sources of dynamic loads [5, 7-9] may include; (i) seismic disturbance (ii) wind (iii) industrial machinery (iv) human forces (v) moving vehicles (vi) blasting (vii) pile driving (viii) moving load on beam (ix) stationary vibrating loads (x) shock waves (xi) impact and sudden loading.

A dynamic load [5,6] is not only time-dependent, it is also essentially associated with the common characteristics of inducing or developing vibrations on the structure it acts. Such vibrations can often get excited at the same frequency as that of the forcing external dynamic load. Should the frequency of the external force coincide with any of the natural frequencies of the frame, a phenomenon known as resonance would ensue, leading to very large deflections and stresses beyond what the structure might have been designed for. At this point, the structure can even collapse [5,6,8]. Conventionally, evaluation of the natural frequencies [9], is a matter of priority interest to the engineer involved in the dynamic analysis of any structural system. This is for the reason the design would be better guided so as to avoid a situation where the structure ever plunged into the catastrophic regime of resonance amid the knowledge of such natural frequencies.

Shear frame model is a phenomenon of beam stiffness being sufficiently large relative to column stiffness, taken as rationale for assuming beams to be infinite for many practical situations [10]. Shear frame does not permit rotation of its joints but allows lateral vibration motion in its plane on account of the assumed infinite rigidity of its horizontal members [6].

Flaws exist when considering the frame as shear frame. Notable among the flaws are reality that even though shear frame approach can predict the natural frequencies with some degree of accuracy, it does not yield identical results with the actual deformation [11, 12]. Shear frame, as an analysis model [13], is not only insufficient but also deficient as a generalized dynamic model for multi-storey building structure.

Unlike the conventional analysis which assumes the shear frame model, this paper adopts an improved model for dynamic analysis of frames known as: frames with flexible horizontal members [6, 14]. In contrast to shear frames [6, 14], flexibility of beams as well as joint rotations are accounted for in computation of restoring forces which the elements of dynamic analysis are composed.

In numerical study using the improved model [14], it was apparent that, in comparison with the results of shear frame, flexibility of beams and associated joint rotations within a certain range have quite significant influence on model frequency of multi-storey building frames.

The double-bay four-storey simulated building frame, without any loss of generality, is considered as a multi-degrees of freedom (MDOF) system which permits only lateral translation or sway motion in its plane, fig, 4.

This paper presents the evaluation of the natural frequencies of an undamped vibrating double-bay four-storey frame with stiffened joints fig.3. This is with the aim to investigate the effect of stiffened joints on the natural frequencies using the dynamic displacement or stiffness method of analysis.

This paper seeks to achieve the aim through the following objectives.

- i. To determine the bending moments due to unit translation at the four floor levels;

- ii. To evaluate the shear forces due to unit translation at the four floor levels;
- iii. To determine the restoring forces;
- iv. To evaluate the natural frequencies of the simulated normal rigid frame and stiffened frames;

2. STRUCTURAL MODEL AND DYNAMIC FLOOR MASS

In this study, the building configuration adopted is simple and regular. It can essentially serve as an office or classroom block having a plan as shown in Fig. 1. A reinforced concrete frame of four-floors simulated model shown in Fig. 2 is to be used as the case study. Handling procedural arrangement is in four major phases, each comprising of four case studies as stated thus:

Phase 1 treats the same frame as a normal rigid frame structure under four separate cases, namely; unit translation at: (i) fourth floor level (ii) third floor level (iii) second floor level and (iv) first floor level respectively. Phase 2 deals with the simulated frame as a stiffened frame of 250mm stiffening length, for similar cases of translation as in phase 1. Phases 3 and 4 are similar to phase 2 but for the stiffening lengths are increased to 400mm and 750mm respectively. The floor thickness for each floor level is 150mm but for the topmost floor level, which is 125mm thick. A concrete grade of 30N/mm² and of density 24KN/m³ is adopted. A provision of floor finishes of IKN/M² is made in respect of each of the floors. These culminated into the numerical value of dynamic floor mass, $M_1=M_2=..... = M_{n-1}= 34404\text{Kg}$ and $M_n=27523\text{Kg}$; where n is taken to be number of floors.

2.1 DYNAMIC MODEL

Although every structural system is essentially system with distributed mass i.e. system is continuous and possesses infinite number of degrees of freedom (15, 16), the lumped mass system adopted in this work for the dynamic analysis provides satisfactorily approximate results (17).

Therefore, in the course of this work, each of the simulated double-bay, multi-storey frames is modeled as a structure having finite number of degrees of freedom, assuming lumped mass element concentration at the right corner of each floor level as shown in Fig. 4.

2.2 DYNAMIC ANALYSIS

The determination of the natural frequencies associated with the free vibration of structural system [9, 18] constitutes the basic principle in the dynamic analysis of structural system. As resonance at the lowest frequency would result in the maximum effects, [19], it is only the basic tone of vibration that is usually considered to be of paramount importance. [20] Indicated that with increasing number of floors, ten and above, the flexibility of building structures increases so as to necessitate inclusion of the higher modal frequencies determination as well, even if the buildings are regular. Although the number of floors for the building structure in this work did not fall within the stipulated in [20], the work did not deem it unnecessary to include determination of higher modal frequencies since resonance at the higher modal frequencies does not result in no dynamic effects.

2.3 EQUATION OF MOTION

Although the equations of static equilibrium differ from equations of dynamic equilibrium, D 'Alembert proposed a principle that makes it possible to use equation of static equilibrium in solving dynamic problems. This principle introduces terms that take into cognizance inertia forces expressed as products of masses by accelerations i.e. by the second derivatives of linear or angular displacements over time. The introduction of such terms (14, 1) automatically transforms equations of static equilibrium into equations of dynamic equilibrium; thus:

$$M \ddot{x}_{(t)} + C \dot{x}_{(t)} + K x_{(t)} = P_{(t)} \dots\dots\dots (1)$$

Where: M, C, K are matrices of mass elements, damping elements and stiffness coefficients respectively:

$P_{(t)}$ is the column vector representing the external excitations.

$X_{(t)}, \dot{X}_{(t)}, \ddot{X}_{(t)}$ are vectors of displacements, velocities and accelerations of a system having finite number of degrees of freedom. Since this work focuses on the regime of free vibration only, there is complete absence of external excitation for this system. Therefore, the exciting force vector is assumed to be Zero. In this case, equation (1) becomes:

$$M \ddot{x}_{(t)} + C \dot{x}_{(t)} + Kx_{(t)} = 0 \dots\dots\dots (2)$$

Generally, to a certain degree, all vibrating systems are subject to damping effect for the reason energy is dissipated by friction and other resistances. Nonetheless, in some cases, damping is very small or the dynamic disturbance on the vibrating system, such as building, act for relatively short duration in such a manner the effect of damping becomes very, negligible and quite unimportant [21, 22, 23]. Hence, this work completely neglects the effect of damping so equation (2) reduces to:

$$M \ddot{x}_{(t)} + Kx_{(t)} = 0 \dots\dots\dots (3)$$

Adopting the stiffness matrix and displacement approach gives the impetus for evaluating the natural frequencies of the simulated frames in the way as solving the non-trivial equation thus:

$$\left| M^{-1} K_s - W^2 I \right| = 0 \dots\dots\dots (4)$$

Where:

- $M^{-1} K_s =$ dynamic matrix
 and
 M^{-1} = inverse matrix of mass system
 K_s = stiffness matrix
 W = eigenvalues
 I = identity matrix

3. METHODOLOGY

For frames with normal rigid joints, the traditional rotational and translational stiffness coefficients suitable for frames with normal rigid joints [6] are used for members and deemed to be fixed at both ends of the member where:

Case I: Rotational Stiffness Coefficients

$$M_{(o)} = \frac{4EI}{L}, \quad M_{(L)} = \frac{-EI}{L} \dots\dots\dots 5$$

$$Q_{(o)} = \frac{-6EI}{L^2}, \quad Q_{(L)} = \frac{-6EI}{L^2}$$

Case II: Translational Stiffness Coefficients

$$M_{(o)} = \frac{6EI}{L^2}, \quad M_{(L)} = \frac{-6EI}{L^2}$$

.....6

$$Q_{(o)} = \frac{-12 EI}{L^3}, \quad Q_{(L)} = \frac{-12 EI}{L^3}$$

The stiffness coefficients for members of frames with stiffened joints are derived following ideas developed by [2], and making adaptation from the equation deduced.

Case I: Rotational Stiffness Coefficients

$$M_A = \frac{4EI}{L} [1 + 3\alpha - 3\alpha^2] \quad \dots\dots\dots(7)$$

$$\alpha = \frac{a}{L}$$

a = length of stiffened A end of the member

L = length of flexible portion of the member

$$Q_A = \frac{-6EI}{L^2} [1 + 2\alpha] \quad \dots\dots\dots(8)$$

$$M_B = \frac{-2EI}{L} [1 + 3(\alpha + \beta) + 6\alpha\beta] \quad \dots\dots\dots(9)$$

$$Q_B = \frac{-6EI}{L^2} [1 + 2\beta] \quad \dots\dots\dots(10)$$

Where:

$$\beta = \frac{b}{L}$$

b = length of stiffened B end of the member.

Note: In this work, a = b in each case. Therefore; $\alpha = \beta$

Case II: Translational Stiffness Coefficients

$$M_A = \frac{6EI}{L} [1 + 2\alpha] \quad \dots\dots\dots(11)$$

$$Q_A = Q_B = \frac{-12 EI}{L^3} \quad \dots\dots\dots(12)$$

$$M_B = \frac{-6EI}{L^2} [1 + 2\beta] \quad \dots\dots\dots(13)$$

i. The bending moment values for each of the member ends is computed as: $M = M_p + \sum_{i=1}^n M_i X_i$ (14)

ii. Where M_p is replaced with $R_i \Delta_j$ since there is no external loading but unit translation, i.e. effects of applied unit translation at various floor levels, $R_i \Delta_j$, constitute the load vectors for the equilibrium equations.

iii. The final shear forces on each of the members of the frame is corrupted from the corresponding bending moments as:

$$H_i = \frac{M 'e' - M 'b' }{L} \dots\dots\dots(15)$$

Where:

$M 'e'$ = value of moment at the far right end of the member length

$M 'b'$ = value of the moment at the left end of the member length

L = length of member

iv. The restoring forces, k_{ij} , are systematically computed from appropriate shear force diagram as:

$$K_{ij} = \sum_{k=i}^n H_q \dots\dots\dots(16)$$

Where: H_q = shear force value,
 $q = 1, 2, \dots, n$

v. The natural frequencies of the simulated building frame with values of $a = b = 0$, are evaluated using the method of classical displacement with respect to frames with flexible horizontal members model [FHMM].

In addition, the natural frequencies of the simulated building frames with values of $a = b = 250\text{mm}$, $a = b = 400\text{mm}$ and $a = b = 750\text{mm}$, respectively, on separate notes, are evaluated using the same method of classical displacement with respect to frames with flexible horizontal members model as well.

4. RESULTS AND DISCUSSIONS

Results

Tables 1 through 8 compare the bending moment and shear force resistances produced from the simulated frame with normal rigid joints and simulated frames with stiffened joints.

i. The results revealed: (i) both the moment and shear force resistances from the stiffened frames were quite higher than the corresponding values from the normal rigid frame, no matter the lengths of stiffening.

ii. The values of bending moment and shear force resistances increased with the length of stiffening.

iii. It would be observed that in each phase, table 9, restoring force, k_{ij} , values satisfied Maxwell, Law of reciprocal.

iv. It is observed the natural frequencies produced in the frame with stiffened joints were greater than the corresponding values of natural frequencies for the normal rigid frame, table 10.

v. In addition, natural frequencies increased with the length of stiffening, table 10.

vi. Natural frequency increased with stiffening factor, fig. 5.

Discussion

The observed results could infer the following:-

- i. The results were in consonance with earlier studies of Osadebe [2] and Aminu [3] regarding enhanced performance of frames with stiffened joints. This is so because, since bending moment and shear force resistances are significantly enhanced in the frames with stiffened joints, displacements associated with them such as joint rotation and joint rotation or side sway, flexural and shear cracks, would correspondingly reduce thereby leading to increased stability, durability and enhanced structural integrity of the frame structure.
- ii. The result suggests an optimal length of stiffening, say L_o , exists at which the maximum benefits of stiffening is utilized, i.e. at L_o , a balance would be hit between the desire for bending moment reduction and increased natural frequencies. Hence, it would be likely that above L_o , the gains of stiffening in terms of bending moment reduction and stability of frame would commence their depreciation, while below L_o , the framed structure exhibits traits of normal rigid frame in respect of low natural or modal frequencies.
- iii. Stiffening of joints of framed structures increases their rigidity against dynamic forces since natural frequencies get enhanced through stiffening. Therefore, with stiffened joints vulnerability of the framed structure to resonance with any of the dynamic forces would be credibly reduced.

5. CONCLUSION AND RECOMMENDATION

CONCLUSION

Based on those results of this investigation, it is concluded that stiffening of connecting joints of framed structures increases the natural frequencies of the framed structure while at the same time satisfying the stability requirement.

Therefore, stiffening the joints of multi-storey buildings would safeguard these buildings against resonance and its associated repercussions and consequences.

An area to be investigated further would be the functional relationship between bending moment (m) natural or modal frequencies (w), and optimal length of stiffening, L_o .

RECOMMENDATION

This study presents the following recommendations drawn basically from this investigation:

- i. Structural dynamics analysts and designers of frame structures should always include determination of not only values of the lowest modal frequencies but also those of the higher modal frequencies because resonance at the higher modal frequencies does not result in zero dynamic effects.
- ii. Stiffening of joints of frame structures should be given credible encouragement as much as possible.
- iii. Static and dynamic analysts and designers of frame structures should pursue further the exploitation of the gains of joints stiffening for enhanced structural integrity of such structures.

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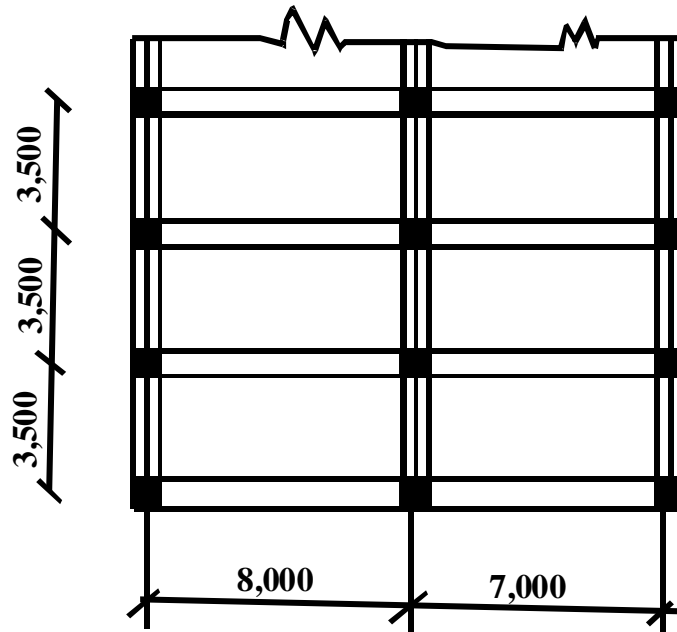


Fig. 1: Plan of the simulated double-bay Multi-Storey Framed Building Structure

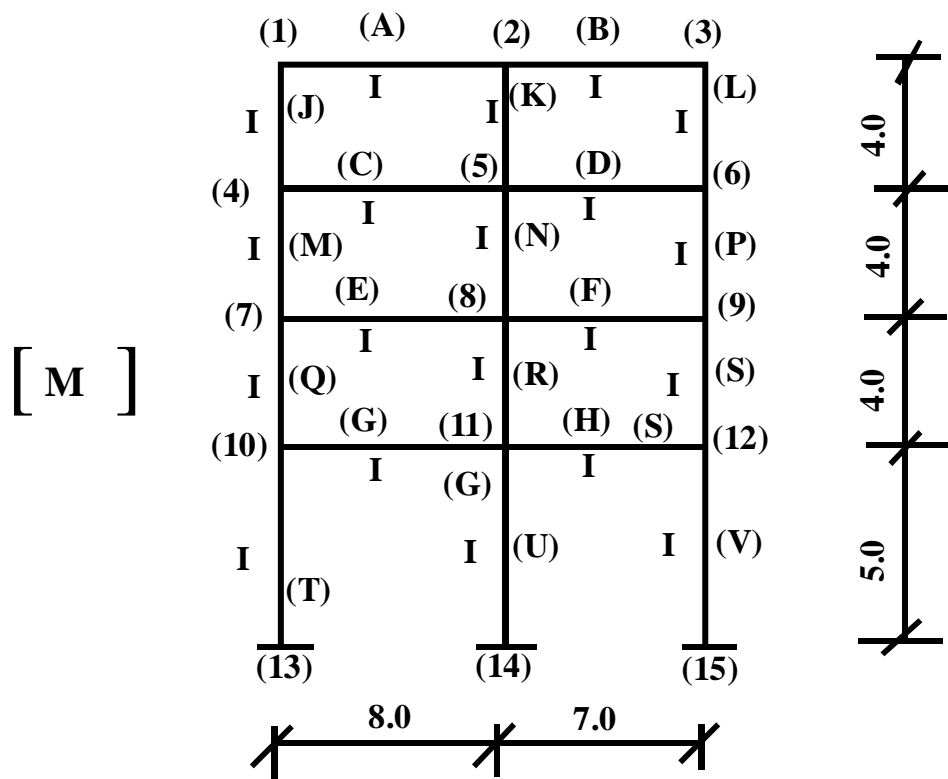


Fig. 2: A Double-bay Four-Storey simulated Normal Rigid Building Frame.

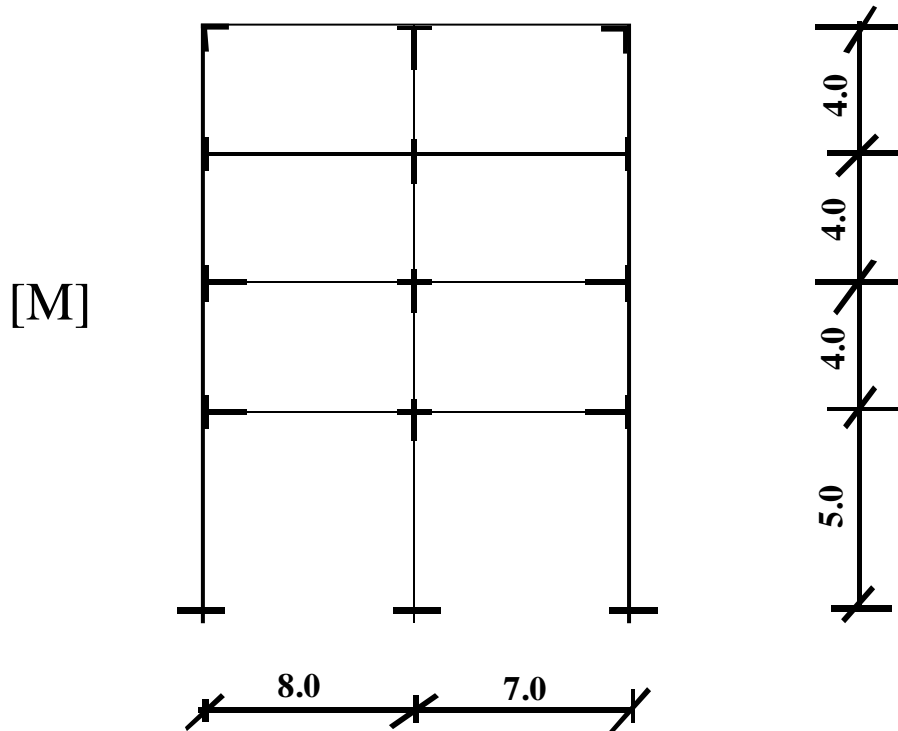


Fig. 3: Double-bay Four-storey simulated Stiffened Building Frame

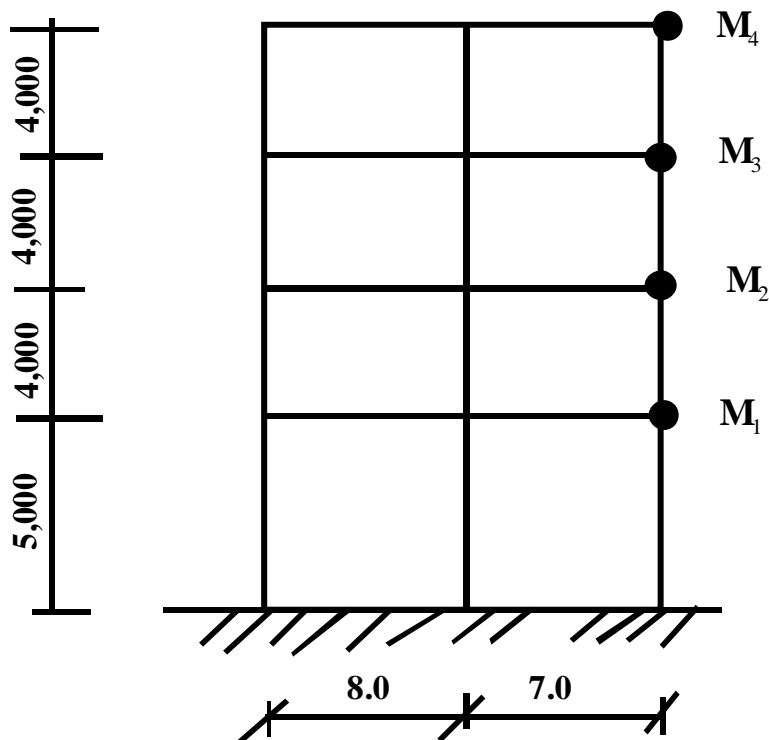


Fig. 4: Dynamic model for the Double-bay four-storey simulated Building Frame

Table 1: Values of bending Moment due to unit translation at 4th floor level only.

	Normal Rigid Frame a = b = 0	Stiffened Frame a = b = 250mm (EI)	Stiffened Frame a = b = 400mm (EI)	Stiffened Frame a = b = 750mm (EI)
M ₁₋₂	0.12574	0.16790	0.20424	0.35143
M ₁₋₄	0.12574	0.16790	0.20424	0.35143
M ₂₋₁	-0.10436	-0.14337	-0.17719	-0.31205
M ₂₋₃	0.11622	0.16368	0.20548	0.37621
M ₂₋₅	-0.22058	-0.30705	-0.38268	-0.68827
M ₃₋₂	-0.13759	-0.18785	-0.23177	-0.41246
M ₃₋₆	-0.13759	-0.18785	-0.23177	-0.41246
M ₄₋₁	-0.17070	-0.22532	-0.26624	0.33908
M ₄₋₅	0.07500	0.09648	0.11445	0.18634
M ₄₋₇	-0.09570	-0.12884	-0.15179	-0.15273
M ₅₋₄	-0.07033	-0.09290	-0.11200	-0.19219
M ₅₋₂	0.23213	0.32174	0.39796	0.68711
M ₅₋₆	0.07971	0.10837	0.13313	0.24013
M ₅₋₈	0.08210	0.12045	0.15283	0.25477
M ₆₋₃	0.17838	0.23928	0.28686	0.40169
M ₆₋₅	-0.08438	-0.11185	-0.13540	-0.23389
M ₆₋₉	0.09400	0.12743	0.15148	0.16779
M ₇₋₄	0.03206	0.05397	0.07523	0.16539
M ₇₋₈	-0.01324	-0.02169	-0.03038	-0.08394
M ₇₋₁₀	0.01882	0.03226	0.04485	0.08145
M ₈₋₇	0.01070	0.01846	0.02661	0.07978
M ₈₋₅	-0.03289	-0.05951	-0.8806	-0.26784
M ₈₋₉	-0.01187	-0.02104	-0.03087	-0.09646
M ₈₋₁₁	-0.01033	-0.02002	-0.03056	-0.09159
M ₉₋₆	-0.03217	-0.05467	-0.07698	-0.18015
M ₉₋₈	0.01441	0.02422	0.03449	0.09959
M ₉₋₁₂	-0.01776	-0.03046	-0.04247	-0.08057
M ₁₀₋₇	-0.00607	-0.00758	-0.02121	-0.04945
M ₁₀₋₁₁	0.00250	0.00530	0.00878	0.03873
M ₁₀₋₁₃	-0.00356	-0.00758	-0.01241	-0.04945
M ₁₃₋₁₀	0.00178	0.00382	0.00893	0.02633
M ₁₁₋₁₀	-0.00167	-0.00384	-0.00668	-0.03369
M ₁₁₋₈	0.00433	0.01024	0.01807	0.09606
M ₁₁₋₁₂	0.00178	0.00422	0.00748	0.03926
M ₁₁₋₁₄	0.00088	0.00218	0.00393	0.02310
M ₁₄₋₁₁	-0.00044	-0.00110	-0.00283	-0.01230
M ₁₂₋₁₁	-0.00262	-0.00566	-0.00952	-0.04347
M ₁₂₋₉	0.00585	0.01247	0.02061	0.08667
M ₁₂₋₁₅	0.00323	0.00681	0.01110	0.04322
M ₁₅₋₁₂	-0.00161	-0.00343	-0.00798	-0.02302

Table 2: Values of bending Moment due to unit translation at 3rd floor level only.

	Normal Rigid Frame a = b = 0	Stiffened Frame a = b = 250mm (EI)	Stiffened Frame a = b = 400mm (EI)	Stiffened Frame a = b = 750mm (EI)
M ₁₋₂	-0.14847	-0.20655	-0.25976	-0.52565
M ₁₋₄	-0.14847	-0.20655	-0.25976	-0.52565
M ₂₋₁	0.12004	0.17146	0.21867	0.43660
M ₂₋₃	-0.13313	-0.19467	-0.25186	-0.51706
M ₂₋₅	0.25316	0.36614	0.47051	0.95368
M ₃₋₂	0.16156	0.22926	0.29184	0.60009
M ₃₋₆	0.16156	0.22926	0.29184	0.60009
M ₄₋₁	0.27573	0.36947	0.43773	0.48297
M ₄₋₅	0.00949	0.01695	0.02513	0.09028
M ₄₋₇	0.28522	0.38643	0.46287	0.57324
M ₅₋₄	-0.00498	-0.00964	-0.01486	-0.04478
M ₅₋₂	-0.31455	-0.44262	-0.55100	-0.94431
M ₅₋₆	0.00504	0.00996	0.01550	0.04439
M ₅₋₈	-0.32456	-0.46221	-0.58134	-1.03348
M ₆₋₃	-0.28059	-0.37998	-0.45524	-0.56203
M ₆₋₅	-0.00955	-0.01720	-0.02557	-0.08802
M ₆₋₉	-0.29014	-0.39718	-0.48082	-0.65005
M ₇₋₄	-0.22344	-0.31539	-0.39222	-0.57887
M ₇₋₈	0.09602	0.13139	0.16344	0.30406
M ₇₋₁₀	-0.12742	-0.18399	-0.22877	-0.27481
M ₈₋₇	-0.08538	-0.11948	-0.15073	-0.30395
M ₈₋₅	0.27505	0.40055	0.51530	1.05038
M ₈₋₉	0.09605	0.13799	0.17703	0.37445
M ₈₋₁₁	0.09362	0.14307	0.18754	0.37188
M ₉₋₆	0.22989	0.32763	0.41122	0.65651
M ₉₋₈	-0.10670	-0.14968	-0.18925	-0.37241
M ₉₋₁₂	0.12319	0.17797	0.22196	0.28412
M ₁₀₋₇	0.04150	0.07411	0.10901	0.29675
M ₁₀₋₁₁	-0.01781	-0.03164	-0.04669	-0.13821
M ₁₀₋₁₃	0.02369	0.04247	0.06231	0.15853
M ₁₃₋₁₀	-0.01184	-0.02139	-0.04480	-0.08431
M ₁₁₋₁₀	0.01342	0.02498	0.05412	0.12871
M ₁₁₋₈	-0.03778	-0.07103	-0.10846	-0.39145
M ₁₁₋₁₂	-0.01472	-0.02805	-0.04330	-0.15389
M ₁₁₋₁₄	-0.00964	-0.01799	-0.02719	-0.10886
M ₁₄₋₁₁	0.00482	0.00906	0.01955	0.05797
M ₁₂₋₁₁	0.01911	0.03460	0.05175	0.16117
M ₁₂₋₉	-0.04104	-0.07356	-0.10862	-0.30487
M ₁₂₋₁₅	-0.02193	-0.03896	-0.05687	-0.14369
M ₁₅₋₁₂	0.01097	0.01962	0.04089	0.07652

Table 3: Values of bending Moment due to unit translation at 2nd floor level only.

	Normal Rigid Frame a = b = 0	Stiffened Frame a = b = 250mm (EI)	Stiffened Frame a = b = 400mm (EI)	Stiffened Frame a = b = 750mm (EI)
M ₁₋₂	0.02694	0.04770	0.07082	0.26946
M ₁₋₄	0.02694	0.04770	0.07082	0.26946
M ₂₋₁	-0.01797	-0.03345	-0.05089	-0.17105
M ₂₋₃	0.01925	0.03659	0.05631	0.18666
M ₂₋₅	-0.03722	-0.07005	-0.10721	-0.35770
M ₃₋₂	-0.02823	-0.05067	-0.07577	-0.28018
M ₃₋₆	-0.02823	-0.05067	-0.07577	-0.28018
M ₄₋₁	-0.12571	-0.18024	-0.22225	-0.21830
M ₄₋₅	-0.09944	-0.13888	-0.17648	-0.41707
M ₄₋₇	-0.22515	-0.31914	-0.39874	-0.63538
M ₅₋₄	0.08663	0.12252	0.15621	0.32955
M ₅₋₂	0.09243	0.14018	0.18233	0.35256
M ₅₋₆	-0.09718	-0.14075	-0.18197	-0.38753
M ₅₋₈	0.27624	0.40345	0.52051	1.06964
M ₆₋₃	0.12155	0.17437	0.21576	0.23506
M ₆₋₅	0.10998	0.15685	0.20166	0.47063
M ₆₋₉	0.23153	0.33123	0.41742	0.70559
M ₇₋₄	0.29978	0.41283	0.50107	0.60058
M ₇₋₈	-0.00036	-0.00089	-0.00100	-0.04097
M ₇₋₁₀	0.29942	0.41194	0.50005	0.64150
M ₈₋₇	0.00043	0.00117	0.00193	0.00243
M ₈₋₅	-0.32512	-0.46320	-0.58259	-1.06722
M ₈₋₉	-0.00050	-0.00142	-0.00248	-0.01162
M ₈₋₁₁	-0.32418	-0.46060	-0.57818	-1.05315
M ₉₋₆	-0.30295	-0.42008	-0.51367	-0.67472
M ₉₋₈	0.00043	0.00113	0.00154	0.03099
M ₉₋₁₂	-0.30252	-0.41895	-0.51212	-0.70571
M ₁₀₋₇	0.22326	-0.31600	-0.39802	-0.65050
M ₁₀₋₁₁	0.10171	0.14342	0.18137	0.32764
M ₁₀₋₁₃	-0.12155	-0.17257	-0.21665	-0.32286
M ₁₃₋₁₀	0.06077	0.08691	0.15579	0.17193
M ₁₁₋₁₀	-0.08947	-0.12854	-0.16457	-0.33017
M ₁₁₋₈	0.27236	0.39441	0.50627	1.07352
M ₁₁₋₁₂	0.10051	0.14807	0.19264	0.40725
M ₁₁₋₁₄	0.08238	0.11781	0.14904	0.33612
M ₁₄₋₁₁	-0.04119	-0.05933	-0.10717	-0.17899
M ₁₂₋₁₁	-0.11275	-0.16269	-0.20886	-0.40241
M ₁₂₋₉	0.22940	0.32733	0.41489	0.71534
M ₁₂₋₁₅	0.11665	0.16464	0.20602	0.31293
M ₁₅₋₁₂	0.05833	-0.08291	-0.14815	-0.16664

Table 4: Values of bending Moment due to unit translation at 1st floor level only

	Normal Rigid Frame a = b = 0	Stiffened Frame a = b = 250mm (EI)	Stiffened Frame a = b = 400mm (EI)	Stiffened Frame a = b = 750mm (EI)
M ₁₋₂	-0.00489	-0.01075	-0.01848	-0.12792
M ₁₋₄	-0.00489	-0.01075	-0.01848	-0.12792
M ₂₋₁	0.00254	0.00604	0.01075	0.05390
M ₂₋₃	-0.00257	-0.00622	-0.01115	-0.04957
M ₂₋₅	0.00511	0.01226	0.02189	0.10347
M ₃₋₂	0.00492	0.01088	0.01874	0.12097
M ₃₋₆	0.00492	0.01088	0.01874	0.12097
M ₄₋₁	0.02427	0.04346	0.06232	0.09824
M ₄₋₅	0.01723	0.02993	0.04407	0.18448
M ₄₋₇	0.04151	0.07339	0.10638	0.28272
M ₅₋₄	-0.01265	-0.02268	-0.03372	-0.10984
M ₅₋₂	-0.01061	-0.02060	-0.03141	-0.11058
M ₅₋₆	0.01379	0.02525	0.03795	0.11828
M ₅₋₈	-0.03705	-0.06853	-0.10308	-0.33869
M ₆₋₃	-0.02256	-0.04019	-0.05754	-0.09574
M ₆₋₅	-0.01839	-0.03240	-0.04803	-0.18983
M ₆₋₉	-0.04095	-0.07259	-0.10557	-0.28548
M ₇₋₄	-0.12667	-0.18129	-0.22457	-0.24392
M ₇₋₈	-0.09450	-0.12711	-0.15644	-0.33424
M ₇₋₁₀	-0.22117	-0.30838	-0.38102	-0.57808
M ₈₋₇	-0.09332	-0.13089	-0.16456	-0.31517
M ₈₋₅	0.26661	0.37876	0.47719	0.91701
M ₈₋₉	0.09019	0.13422	0.17189	0.33504
M ₈₋₁₁	0.08309	0.11366	0.14077	0.26680
M ₉₋₆	0.12211	0.17445	0.21626	0.25108
M ₉₋₈	0.10474	0.14412	0.17976	0.37936
M ₉₋₁₂	0.22685	0.31858	0.39602	0.63044
M ₁₀₋₇	0.27918	0.37118	0.44331	0.55637
M ₁₀₋₁₁	-0.01902	-0.03730	-0.05387	-0.08830
M ₁₀₋₁₃	0.26016	0.33388	0.38944	0.46813
M ₁₃₋₁₀	-0.25010	-0.31405	-0.37355	-0.45937
M ₁₁₋₁₀	0.01914	0.03720	0.05412	0.12129
M ₁₁₋₈	-0.30155	-0.41434	-0.50924	-0.92016
M ₁₁₋₁₂	-0.02188	-0.04365	-0.06467	-0.15820
M ₁₁₋₁₄	-0.26053	-0.33350	-0.39046	-0.64067
M ₁₄₋₁₁	0.25027	0.31386	0.36219	0.55127
M ₁₂₋₁₁	0.02177	0.04373	0.06436	0.12585
M ₁₂₋₉	-0.28198	-0.37749	-0.45376	-0.61172
M ₁₂₋₁₅	-0.26021	-0.33376	-0.38938	-0.48587
M ₁₅₋₁₂	0.25011	0.31398	0.36165	0.46883

Table 5: Values of Shear Force due to unit translation at 4th floor level only

	Normal Rigid Frame a = b = 0	Stiffened Frame a = b = 250mm (EI)	Stiffened Frame a = b = 400mm (EI)	Stiffened Frame a = b = 750mm (EI)
Q ₁₋₂	-0.02876	-0.03891	-0.04768	-0.08294
Q ₂₋₃	-0.03626	-0.05022	-0.06246	-0.11267
Q ₄₋₁	0.07411	0.09831	0.11762	0.17263
Q ₂₋₅	0.11318	0.15720	0.19516	0.34385
Q ₃₋₆	0.07899	0.10678	0.12966	0.20354
Q ₄₋₅	-0.01817	-0.02367	-0.02831	-0.04732
Q ₅₋₆	-0.02344	-0.03146	-0.03836	-0.06772
Q ₇₋₄	-0.03194	-0.04570	-0.05676	-0.07953
Q ₅₋₈	-0.02875	-0.04499	-0.06022	-0.13065
Q ₆₋₉	-0.03154	-0.04553	-0.05712	-0.08699
Q ₇₋₈	0.00299	0.00502	0.00712	0.02047
Q ₈₋₉	0.00375	0.00647	0.00934	0.02801
Q ₁₀₋₇	0.00622	0.01129	0.01652	0.04241
Q ₈₋₁₁	0.00367	0.00757	0.01216	0.04691
Q ₉₋₁₂	0.00590	0.01073	0.01577	0.04181
Q ₁₀₋₁₁	-0.00052	-0.00114	-0.00193	-0.01516
Q ₁₁₋₁₂	-0.00063	-0.00141	-0.00243	-0.01182
Q ₁₃₋₁₀	-0.00107	-0.00228	-0.00427	-0.01516
Q ₁₁₋₁₄	-0.00026	-0.00066	-0.00135	-0.00708
Q ₁₂₋₁₅	-0.00097	-0.00205	-0.00382	-0.01325

Table 6: Values of Shear Force due to unit translation at 3rd floor level only

	Normal Rigid Frame a = b = 0	Stiffened Frame a = b = 250mm (EI)	Stiffened Frame a = b = 400mm (EI)	Stiffened Frame a = b = 750mm (EI)
Q ₁₋₂	0.03356	0.04725	0.05980	0.12028
Q ₂₋₃	0.04210	0.06056	0.07769	0.15959
Q ₄₋₁	-0.10605	-0.14401	-0.17437	-0.25216
Q ₂₋₅	-0.14193	-0.20219	-0.25538	-0.47450
Q ₃₋₆	-0.11054	-0.15231	-0.18677	-0.29053
Q ₄₋₅	-0.00181	-0.00332	-0.00500	-0.01688

Q ₅₋₆	-0.00208	-0.00388	-0.00587	-0.01892
Q ₇₋₄	0.12717	0.17546	0.21377	0.28803
Q ₅₋₈	0.14990	0.21569	0.27416	0.52097
Q ₆₋₉	0.13001	0.18120	0.22301	0.32664
Q ₇₋₈	-0.02268	-0.03136	-0.03927	-0.09297
Q ₈₋₉	-0.02896	-0.04110	-0.05233	-0.10669
Q ₁₀₋₇	-0.04223	-0.06453	-0.08445	-0.14289
Q ₈₋₁₁	-0.03285	-0.05353	-0.07400	-0.09297
Q ₉₋₁₂	-0.04106	-0.04106	-0.08265	-0.14725
Q ₁₀₋₁₁	0.00390	0.00708	0.01260	0.03337
Q ₁₁₋₁₂	0.00483	0.00895	0.01358	0.04501
Q ₁₃₋₁₀	0.00711	0.01277	0.02142	0.04857
Q ₁₁₋₁₄	0.00289	0.00541	0.00935	0.03337
Q ₁₂₋₁₅	0.00658	0.01172	0.01955	0.04404

Table 7: Values of Shear Force due to unit translation at 2nd floor level only.

	Normal Rigid Frame a = b = 0	Stiffened Frame a = b = 250mm (EI)	Stiffened Frame a = b = 400mm (EI)	Stiffened Frame a = b = 750mm (EI)
Q ₁₋₂	-0.00561	-0.01014	-0.01521	-0.05506
Q ₂₋₃	-0.00678	-0.01247	-0.01887	-0.06669
Q ₄₋₁	0.03816	0.05699	0.07327	0.12194
Q ₂₋₅	0.03241	0.05256	0.07239	0.17757
Q ₃₋₆	0.03745	0.05626	0.07239	0.12881
Q ₄₋₅	0.02326	0.03268	0.04159	0.09333
Q ₅₋₆	0.02959	0.04251	0.05480	0.12259
Q ₇₋₄	-0.13123	-0.18299	-0.22495	-0.30899
Q ₅₋₈	-0.15034	-0.21666	-0.27578	-0.53422
Q ₆₋₉	-0.13362	-0.18783	-0.23277	-0.34508
Q ₇₋₈	0.00010	0.00026	0.00037	0.000543
Q ₈₋₉	0.00013	0.00036	0.00057	0.00609
Q ₁₀₋₇	0.13067	0.18199	0.22452	0.32300
Q ₈₋₁₁	0.14914	0.21375	0.27111	0.53167
Q ₉₋₁₂	0.13298	0.18657	0.23175	0.35526
Q ₁₀₋₁₁	-0.02390	-0.03400	-0.04324	-0.08223

Q ₁₁₋₁₂	-0.03047	-0.04439	-0.05736	-0.11567
Q ₁₃₋₁₀	-0.03646	-0.05190	-0.07449	-0.09896
Q ₁₁₋₁₄	-0.10216	-0.03543	-0.05124	-0.10302
Q ₁₂₋₁₅	-0.10206	-0.04951	-0.07083	-0.09591

Table 8: Values of shear force due to Unit Translation at 1st Floor Level only

	Normal Rigid Frame a = b = 0	Stiffened Frame a = b = 250mm (EI)	Stiffened Frame a = b = 400mm (EI)	Stiffened Frame a = b = 750mm (EI)
Q ₁₋₂	0.00093	0.00210	0.00365	0.02273
Q ₂₋₃	0.00107	0.00244	0.00427	0.02436
Q ₄₋₁	-0.00729	-0.01355	-0.02020	-0.05654
Q ₂₋₅	-0.00393	-0.00822	-0.01333	-0.05351
Q ₃₋₆	-0.00687	-0.01277	-0.01907	-0.05418
Q ₄₋₅	0.00374	-0.00658	-0.00972	-0.03679
Q ₅₋₆	-0.00460	-0.00824	-0.01228	-0.04402
Q ₇₋₄	0.04205	0.06367	0.08274	0.13166
Q ₅₋₈	0.03181	0.05069	0.06874	0.16843
Q ₆₋₉	0.00408	0.06176	0.08046	0.13414
Q ₇₋₈	0.02220	0.03010	0.03715	0.07513
Q ₈₋₉	0.02829	0.03929	0.04919	0.09922
Q ₁₀₋₇	-0.12509	-0.16989	-0.20608	-0.28361
Q ₈₋₁₁	-0.14204	-0.19828	-0.24661	-0.45929
Q ₉₋₁₂	-0.12721	-0.17402	-0.21245	-0.31054
Q ₁₀₋₁₁	0.00477	0.00931	0.01350	0.02620
Q ₁₁₋₁₂	0.00624	0.01248	0.01843	0.04058
Q ₁₃₋₁₀	0.10205	0.12959	0.15260	0.18550
Q ₁₁₋₁₄	0.10216	0.12947	0.15053	0.23839
Q ₁₂₋₁₅	0.10206	0.12955	0.15021	0.19094

Table 9: Values of Restoring Forces, K_{ij} , for the four simulated frames

K_{ij}	Normal Rigid Frame $a = b = 0$	Stiffened Frame $a = b = 250\text{mm}$ (EI)	Stiffened Frame $a = b = 400\text{mm}$ (EI)	Stiffened Frame $a = b = 750\text{mm}$ (EI)
K_{11}	0.70061	0.93079	1.11845	1.66827
K_{21}	-0.50896	-0.72872	-0.91051	-1.49775
K_{31}	0.13271	0.21074	0.28797	0.60271
K_{41}	-0.01809	-0.03455	-0.05324	-0.16543
K_{12}	0.50896	-0.71872	-0.91051	-1.49775
K_{22}	0.82798	1.16979	1.46088	2.39821
K_{32}	-0.52321	-0.75329	-0.95204	-1.61660
K_{42}	0.10802	0.16580	0.21854	0.42831
K_{13}	0.13271	0.21074	0.28797	0.60271
K_{23}	-0.52321	-0.75329	-0.95204	-1.61660
K_{33}	0.76559	1.07086	1.32746	2.15282
K_{43}	-0.35851	-0.49851	-0.61653	-1.01719
K_{14}	-0.01809	-0.03455	-0.05324	-0.16543
K_{24}	0.10802	0.16580	0.21854	0.42831
K_{34}	-0.35851	-0.49851	-0.61653	-1.01719
K_{44}	0.26628	0.36228	0.44244	0.72002

Table 10: Values of Natural Frequencies, w_i , for the four simulated Building Frames

Natural frequencies W_i	Normal Rigid Frame $a = b = 0$	Stiffened Frame $a = b = 250\text{mm}$	Stiffened Frame $a = b = 400\text{mm}$	Stiffened Frame $a = b = 750\text{mm}$
	$(\sqrt{E I_x} 10^{-3} \text{ rad/se})$	$(\sqrt{E I_x} 10^{-3} \text{ rad/se})$	$(\sqrt{E I_x} 10^{-3} \text{ rad/se})$	$(\sqrt{E I_x} 10^{-3} \text{ rad/se})$
	$\alpha = \beta = 0$	$\alpha = \beta = 0.04$	$\alpha = \beta = 0.06$	$\alpha = \beta = 0.13$
W_1	0.65672	0.81697	0.93730	1.17533
W_2	2.13741	2.65992	3.12890	3.76203
W_3	3.89275	4.97313	6.08300	7.26325
W_4	5.58564	7.58335	9.75407	12.63431

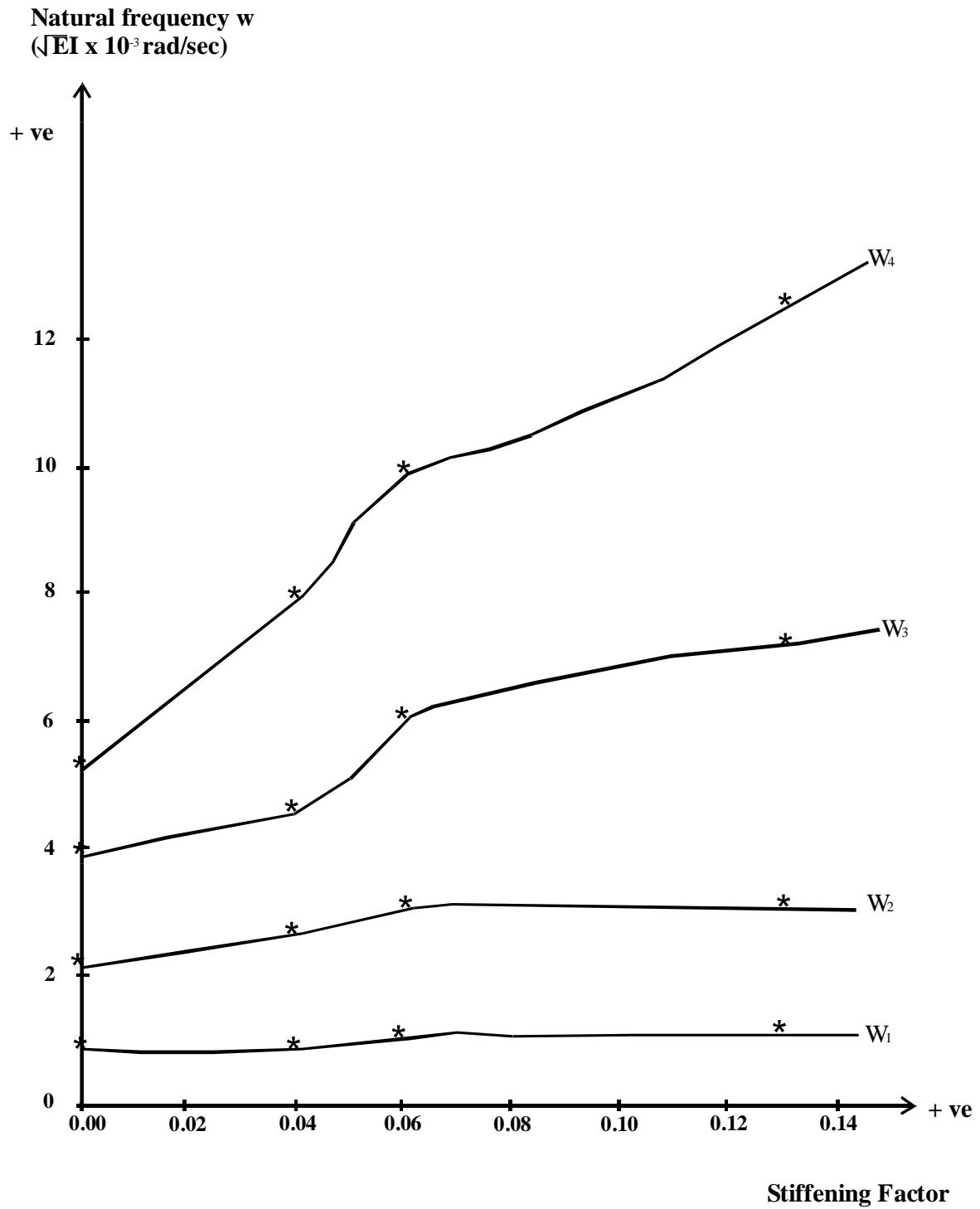


Fig 5: Graph of Natural Frequencies versus Stiffening Factors, α and β .